

Cycle Ambiguity Resolution in GNSS-R Carrier Phase Altimetry

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• Coherent reflections have been collected from space over sea ice and ocean (at grazing geometry as that for RO):

\rightarrow CHAMP, UK-DMC, TDS-1, CyGNSS ...

- ESA has recently studied, assessed or approved GNSS Reflectometry mission concepts based partly or fully on carrier phase observables for altimetry applications:
 - \rightarrow **GEROS-ISS** on board the International Space Station
 - \rightarrow **G-TERN** Earth Explorer 9 proposal
 - \rightarrow FSSCat cubesat mission (approved)
 - \rightarrow SaaS Pioneer cubesat mission (approved)
 - → **PRETTY** cubesat mission (<u>approved</u>)
- This presentation addresses the issue of carrier phase ambiguity resolution

Carrier Phase Ambiguity



• Carrier phase observable

$$\lambda l = r + a - e + \lambda W_s - \lambda n$$

- λ wavelength
- *l* measured relative phase (cycles)
- *r* relative distance
- *a* relative tropospheric path excess
- *e* relative ionospheric phase path advance
- W_s wind-up due to reflection at specular point S (cycles)
- *n* relative integer ambiguity (cycles)

Carrier Phase Ambiguity



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• Carrier phase observable (interferometric processing)

$$\lambda l = r + a - e + \lambda W_s - \lambda n$$

- The quantity of interest is **r**
- To find *r* it is necessary:

→ to estimate a, e and (λW_s) to well less than λ → then to solve for the integer ambiguity n

• <u>Requisite</u>: position of receiver and GNSS transmitter known to better than λ (only relative vector projected errors matter)

Considerations on Geometry



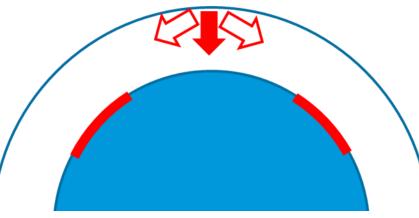


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Grazing Geometry



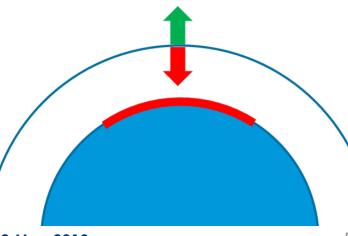
- Observable targets: **all surfaces** (Rayleigh criterion)
- Polarization of reflected signal: mixed LHCP/RHCP
- Low integer ambiguity induced by geometry
- Small direct-to-reflected signal angular separation → same antenna
- Integer ambiguity very insensitive to instrumental errors
- More pronounced atmospheric effects (larger *a* and *e*)
- Integer ambiguity very insensitive to satellite position errors



Around-Nadir Geometry



- Observable targets: very calm ocean, in-land water, sea ice
- Polarization of reflected signal: LHCP
- Large integer ambiguity in the number of carrier cycles
- Large direct-to-reflected signal angular separation → 2 antennas
- Integer ambiguity **sensitive** to **instrumental errors**
- Lower atmospheric effects (smaller *a* and *e*)
- Integer ambiguity **sensitive** to **receiver satellite radial error**



Tropospheric Excess Path Estimation



- Importantly impacts grazing observations (longer traversed path)
- Dry excess path can be estimated within λ using models
- The variable wet troposphere is problematic for grazing geometry
- Possible way around: solve integer ambiguity at different segments of the entire reflection arc and retain that with lowest residuals

Ionospheric Phase Path



- Affects mostly grazing geometry (longer traversed path)
- The ionospheric phase path advance can be found with dual frequency observations
- Integer ambiguity to be fixed in the part of the arc with lowest TEC
- Trade-off between use of widelane or use of 3 frequencies

Satellite Position Errors



- Position uncertainty of GNSS transmitter not critical as it projects the same along direct and reflected path
- Position uncertainty of GNSS-R satellite not critical either in grazing geometry
- Radial component of position uncertainty of GNSS-R satellite is important in around-nadir geometry
- Position uncertainty of GNSS and GNSS-R can be known to a few centimeters in post-processing thus, not a critical error source
- <u>Note</u>: GNSS-R must carry a dual-frequency POD GNSS receiver

Importance of instrument architecture and processing approach



- Instrument architecture is critically important for the ambiguity resolution, in particular for around-nadir geometry
- Processing approach is fundamental to the quality of the carrier phase observables, both single frequency and widelaning

Integer Ambiguity Resolution



- Many methods exist
- These methods can accommodate widelane and multi-frequency
- Example: the Null Space Method (used in ESA's PROBA-3 mission) uses very few epochs to arrive to a solution
- The Null Space Method is robust against re-initialization (when phase track is only intermittently tracked, or when tracked only for a very short period of time)

Conclusions and Way Forward



- Considerations have been given on ambiguity resolution
- In principle it should be possible to solve for it and retrieve absolute heights from carrier phase GNSS-R altimetry
- A ground-based experiment is proposed for demonstration